### 6.1 Quadrotors

### 6.1.1 Fundamental questions

How much power to hover? Model air as incompressible fluid and analyze velocity and pressure gradients.
Conservation of mass

$$
\rho A v_{i}=\rho A_{2} v_{2}
$$

## Conservation of momentum

$$
T=\left(\rho A v_{i}\right) v_{2}
$$

(mass flow rate x change in velocity)

## Thrust of Pressure Difference

$$
T=A\left(p_{L}-p_{U}\right)
$$

## Induced Power/Ideal Power

$$
P_{\text {induced }}=\frac{T^{3 / 2}}{\sqrt{2 \rho A}}
$$

Propellor "Figure of Merit": Fraction of aerodynamic shaft power converted to useful aerodynamic induced power. Typically $0.3-0.6$.

$$
F M=\frac{P_{\text {induced }}}{P_{\text {shaft }}}
$$

Ground Effect Induced power is lower when hovering near the ground.

### 6.1.2 Demonstrated experiments

Human can produce enough to hover! (npr video)
Gyroplanes going back to 1907.
Coordinated assembly, moving rotors, morphology. Maneuverability. Planes + quadrotors.

### 6.1.3 Equations of control



Figure 6.1: Quadrotor Diagram

$$
\begin{gathered}
F_{i}=k_{F} \sigma_{i}^{2} \\
M_{i}=k_{M} \sigma_{i}^{2} \\
m \ddot{x}=\left[\begin{array}{c}
0 \\
0 \\
-m g
\end{array}\right]+R\left[\begin{array}{c}
0 \\
0 \\
F_{1}+F_{2}+F_{3}+F_{4}
\end{array}\right] \\
\dot{R}=R \hat{\omega} \\
I \dot{\omega}+\omega \times I \omega=\left[\begin{array}{c}
L\left(F_{2}-F_{4}\right) \\
L\left(F_{3}-F_{1}\right) \\
M_{1}-M_{2}+M_{3}-M_{4}
\end{array}\right] \\
{\left[\begin{array}{l}
u_{1} \\
u_{2} \\
u_{3} \\
u_{4}
\end{array}\right]=\left[\begin{array}{ccc}
1 & 1 & 1 \\
0 & L & 1 \\
-L & 0 & -L \\
\gamma & -\gamma & \gamma \\
\hline
\end{array}\right]\left[\begin{array}{l}
k_{P} \sigma_{1}^{2} \\
k_{P} \sigma_{2}^{2} \\
k_{P} \sigma_{3}^{2} \\
k_{P} \sigma_{4}^{2}
\end{array}\right]}
\end{gathered}
$$

Sigmas controlled by motor servos

Abstracted away we get

$$
\begin{gathered}
m \ddot{x}=\left[\begin{array}{c}
0 \\
0 \\
-m g
\end{array}\right]+R\left[\begin{array}{c}
0 \\
0 \\
u_{1}
\end{array}\right] \\
\dot{R}=R \hat{\omega} \\
I \dot{\omega}=-\omega \times I \omega+\left[\begin{array}{c}
u_{2} \\
u_{3} \\
u_{4}
\end{array}\right] \\
R=e^{\hat{z} \psi} e^{\hat{y} \theta} e^{\hat{x} \phi} \\
(\text { yaw, pitch, roll respectively }) \\
\hat{\omega}=R^{T} \dot{R}=\hat{z} \dot{\psi}+e^{-\hat{z} \psi} \hat{y} e^{\hat{z} \psi} \dot{\theta}+e^{-\hat{z} \psi} e^{-\hat{y} \theta} \hat{x} e^{\hat{y} \theta} e^{\hat{z} \psi} \\
{\left[\begin{array}{c}
\dot{\gamma} \\
\dot{\theta} \\
\dot{\phi}
\end{array}\right]=J(\gamma, \theta, \phi) \omega}
\end{gathered}
$$

Call roll, pitch and yaw the outputs $\mathbf{y}$.

$$
\begin{gathered}
{\left[\begin{array}{l}
\ddot{y}_{1} \\
\ddot{y}_{2} \\
\ddot{y}_{3}
\end{array}\right]=\left[\begin{array}{c}
0 \\
0 \\
-m g
\end{array}\right]+R\left[\begin{array}{c}
0 \\
0 \\
u_{1}
\end{array}\right]} \\
\dot{y}_{4}=\dot{\gamma}=\left[\begin{array}{lll}
1 & 0 & 0
\end{array}\right] J \omega
\end{gathered}
$$

Differentiate until the inputs show up

$$
\ddot{y}_{4}=\left[\begin{array}{lll}
1 & 0 & 0
\end{array}\right] \dot{J} \omega+\left[\begin{array}{cccc}
R_{z} & & 0 & \\
0 & a_{42} & a_{43} & a_{44}
\end{array}\right]\left[\begin{array}{l}
u_{1} \\
u_{2} \\
u_{3} \\
u_{4}
\end{array}\right]
$$

Rank 2 matrix! so we can't take the inverse to solve for the input.
Dynamic Extension We keep differentiating the dynamics until the input term is nonzero, then keep differentiating until it has a nonsingular matrix.

On the third derivative of $y$, we get a rank 4 matrix if $u_{1} \neq 0$ and $a_{44} \neq 0$. Can linearize and decouple.


### 6.2 Nonholonomic Systems

## Examples

- twirling a pencil around your fingers! Making and braking contact.
- rolling with the constraint of not slipping
- how to reorient yourself in space

Nonoholonomic mechanics.

### 6.2.1 Pfaffian Constraints

Constraints on the velocities. Given state $q \in \mathcal{R}^{n}$, constraints $i=1, \ldots, k$ of the form

$$
\omega^{i}(q) \dot{q}=0
$$

with $\omega^{i}(q) \in \mathcal{R}^{n}$ are Pfaffian. Assume the rows are linearly indpeendent at $q$ so that the constraints are linearly independent.

Given such constraints, can we convert them into constraints on the states instead?
A single constraint is said to be integrable if $\exists h: \mathcal{R}^{n} \rightarrow \mathcal{R}$ s.t.

$$
\omega^{i}(q) \dot{q}=0 \Longleftrightarrow h(q)=0
$$

From

$$
\frac{\partial^{2} h}{\partial q_{i} \partial q_{j}}=\frac{\partial^{2} h}{\partial q_{j} \partial q_{i}}
$$

we get

$$
\frac{\partial\left(\alpha w_{j}\right.}{\partial q_{i}}=\frac{\partial\left(\alpha w_{i}\right.}{\partial q_{j}}
$$

A set of Pfaffian constraints is holonomic if there exists $h_{i}(q)$ for $i=1, \ldots, k$ such that

$$
\omega^{i}(q) \dot{q}=0 \Leftrightarrow h_{i}(q)=c
$$

It is nonholonomic if there are $p<k$ such constraints, partially nonholonomic if $p>0$ and completely nonholonomic if $p=0$ (there are no such constraints).

Constraints in velocity appearing as constraints in the state. Limiting the state to a ( $n-p$ ) dimensional manifold.

To get the directions we can move we construct the right null space of the constraints. That is

$$
w^{i}(q) g_{j}(q)=0
$$

The allowable trajectories satisfying the Pfaffian constraints are the trajectories of the control system.

$$
\dot{q}=g_{1}(q) u_{1}+\cdots+g_{m}(q) u_{m}
$$

### 6.2.2 Examples

### 6.2.2.1 Raibert's hopper

2 DoF: 1 rotates leg, and 1 extends and retracts.
What is the control law required to make this robot flip in the air?
Pfaffian constraint (angular momentum is conserved)

$$
\begin{gathered}
I \dot{\theta}+m(I+d)^{2}(\dot{\theta}+\dot{\psi})=\left(I+m(I+d)^{2}\right) \dot{\theta}+m(I+d)^{2} \dot{\psi}=0 \\
\dot{q}=\left[\begin{array}{c}
1 \\
0 \\
-\frac{m(I+d)^{2}}{I+m(I+d)^{2}}
\end{array}\right] u_{1}+\left[\begin{array}{l}
0 \\
1 \\
0
\end{array}\right] u_{2}
\end{gathered}
$$

### 6.2.2.2 Planar space robot

Reorient satteliates without using boosters.
Angular momentum constraints on Lagrangian leads to state $q=\left(\psi_{1}, \psi_{2}, \theta\right)^{T}$ having the control system

$$
\dot{q}=\left[\begin{array}{c}
1 \\
0 \\
-\frac{a_{13}}{a_{33}}
\end{array}\right] u_{1}+\left[\begin{array}{c}
0 \\
1 \\
-\frac{a_{23}}{a_{33}}
\end{array}\right] u_{2}
$$

### 6.2.2.3 Rolling without slipping

State $q=(x, y, \theta, \phi)^{T}$

$$
\begin{aligned}
\dot{x}-\rho \cos \theta \dot{\phi} & =0 \\
\dot{y}-\rho \sin \theta \dot{\phi} & =0
\end{aligned}
$$

Control law

$$
\dot{q}=\left[\begin{array}{c}
\rho \cos \theta \\
\rho \sin \theta \\
0 \\
1
\end{array}\right] u_{1}+\left[\begin{array}{l}
0 \\
0 \\
1 \\
0
\end{array}\right] u_{2}
$$

### 6.2.2.4 Front Wheel Drive Car



Figure 6.2: Front Wheel Drive Car

Kinematic model of a car. Steering angle $\phi$, angle of car body is $\theta$, position $x, y$.
Constraints

$$
\begin{aligned}
\sin (\theta+\phi) \dot{x}-\cos (\theta+\phi) \dot{y}-I \cos (\phi) \dot{\theta} & =0 \\
\sin (\theta) \dot{x}-\cos (\theta) \dot{y} & =0
\end{aligned}
$$

Control law

$$
\dot{q}=\left[\begin{array}{c}
\cos \theta \\
\sin \theta \\
\frac{1}{l} \tan \phi \\
0
\end{array}\right] u_{1}+\left[\begin{array}{l}
0 \\
0 \\
0 \\
1
\end{array}\right] u_{2}
$$

### 6.2.2.5 Car with $N$ trailers



Figure 6.3: Car with N Trailers

$$
q=\left(x, y, \phi, \theta_{0}, \ldots, \theta_{N}\right)^{T} \in \mathcal{R}^{N+4}
$$

$N+2$ sets of wheels which roll without slipping gives $N+2$ Pfaffian constraints.

### 6.2.2.6 Firetruck

Rear axle is also steerable (driver in the front and driver in the back). Velocity tangential to the wheels is 0.

